

Aerodynamic Design of Commercial Airplanes – ONLINE (AERO0045)

Instructor: Roelof Vos

Course Description

In this course participants learn how aerodynamics drive the detailed exterior design of transport aircraft. What aerodynamic phenomena play a role in the exterior design of a wing, a cockpit, or an engine intake? What is the effect of aerodynamic add-ons such as vortex generators, fairings, or winglets? What are the advantages and penalties of wing sweep and how can the penalties be mitigated by the aerodynamic design of the wing? Those are the type of questions that are being addressed in this course. Participants learn to understand how the various aircraft components should be shaped in order to fulfill aerodynamic requirements in all corners of the flight envelope. The strong ties between aircraft performance, aircraft aerodynamics, and aircraft exterior design are demonstrated through numerous historical and contemporary examples. Although the main focus is on jet aircraft, the course also covers the effects of propeller installation on the aerodynamic design of the empennage.

Course Highlights

- · Causes for interference drag in high-subsonic conditions
- Effect of Reynolds number on shock-boundary-layer interaction
- Design characteristics of supercritical airfoils
- Mach number effects on flow over multi-element airfoils
- Design of root and tip of swept-wing aircraft
- Stability and control beyond the maximum operating Mach number
- · Propeller slipstream effects on longitudinal stability and yawing moment
- Design constraints resulting from transonic buffet Stalling characteristics of wings with high-lift devices

Who Should Attend?

Designed for aeronautical engineers, pilots with some engineering background, government research laboratory personnel, engineering managers and educators.

Course Outline

Introduction and review of aerodynamic fundamentals:

• Introduction, classification, design requirements and objectives of transport aircraft, certification regulations and design rules, general goals in aerodynamic design

- Boundary-layer aerodynamics and subsonic cruise drag
- Relation between supervelocity and pressure coefficient. Relation between geometry and pressure distribution
- Interference drag and area ruling

Airfoil design:

- Pressure distribution on airfoil sections at high and low speed. Development of the transonic airfoil
- Design of supercritical and sonic airfoils. Reynolds number effects
- Low-speed stalling characteristics of airfoil sections
- Reynolds number and Mach number effects on maximum lift coefficient
- High-speed stall (buffet) on airfoil sections
- Airfoils equipped with high-lift devices: mutual interaction, types of flow separation **Wing design:**
- Wing requirements
- Development of the swept-wing concept: historical background, principle of wing sweep, transonic flow characteristics, effect on boundary layer, effect on maximum lift coefficient, fences
- First generation of swept wing aircraft: tip-stall, aeroelastic deformation
- Pressure distribution on finite swept wings: spanwise drag distribution, forward sweep, fuselage interference, improving velocity distribution at root and tip, minimizing pitching moment
- Design considerations for pressure distributions on finite swept wings
- Discussion of examples of modern swept-wing designs

Empennage design, control surface design and engine integration:

- Horizontal tail surface design: functions and requirements, planform shape, aeroelastic effects, elevator lock
- Vertical tail surface design: functions and requirements, planform shape, effect of dorsal fin
- Control surface design: functions and requirements, linearity, aeroelasticity, stall
- Spoiler panels: functions, effect on pressure distribution, interaction with flaps
- Engine intake design: lip design for take-off and cruise conditions, drag in OEI condition. Exhaust design: effect of ambient pressure, boat tail drag. Thrust reversers: types, interference with empennage

Operation:

- Stalling characteristics of full scale aircraft: stall requirements, mechanism complexity vs. maximum lift coefficient, spanwise stall onset, deep stall
- Take-off performance: requirements, minimum unstick speed, lift-to-drag ratio in takeoff and landing
- The buffet onset boundary: requirements, effect on flight envelope. Flight characteristics between MMO and MD: requirements, Mach tuck, Mach trim compensation

Classroom hours / CEUs

35.00 classroom hours 3.5 CEUs

Certificate Track

Aircraft Design

Course Fees

<u>Early Online Registration fee</u>: \$2,195* <u>Regular Online Registration fee</u>: \$2,395 *Early registration fee is available if you register and pay at least 7 days prior to the course start

Registration is open until the first day of the course; however, early registration is encouraged. The online course fee includes individual access to the Zoom course meetings and to course materials, readings, videos, and resources in Blackboard, the University of Kansas Learning Management System. No additional textbook purchases are required outside the course fee.

U.S. Federal Employee Discount

This course is available to U.S. federal employees at 10% off the registration fee. To receive the federal employee discount, you must enter the code **FGVT116** during the checkout process. Please note that you must validate your eligibility to receive this discount by entering your U.S. government email address (ending in .gov or .mil) when creating your online registration profile. This discount is available for both the early registration and regular registration fees.

Instructor Bio

Roelof Vos is an assistant professor at the Aerospace Engineering Department of Delft University of Technology. He teaches undergraduate courses in conceptual airplane design and two graduate courses on aerodynamic design of transport aircraft and fighter aircraft. He obtained an MSc degree from Delft University of Technology and a Ph.D. degree from The University of Kansas. His research focuses on the development of physics-based analysis methods for the conceptual design of unconventional aircraft and on the assessment of emerging aviation technologies.

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